

Amendments to the Claims:

The following listing of claims will replace all prior versions, and listings, of claims in the application:

1. (Canceled)
2. (Original) A Ni-base superalloy consisting essentially of: by weight %, Co: 9 to 10%, Cr: 9 to 10%, Mo: 0.5 to 1%, W: 6 to 8%, Al: 4 to 5%, Ti: 4 to 5%, Ta: 2 to 3%, Hf: 0.5 to 2.5%, Re: 1 to 3%, C: 0.05 to 0.1%, B: 0.005 to 0.01%, Zr: up to 0.02%, and the balance of Ni and inevitable impurities.
3. (Original) A Ni-base superalloy consisting essentially of: by weight %, Co: 10 to 11%, Cr: 10 to 12%, W: 8 to 9%, Al: 4 to 5%, Ti: 4 to 5%, Nb: up to 1%, Hf: 0.5 to 2.5%, C: 0.05 to 0.15%, B: 0.005 to 0.015%, Zr: 0.01 to 0.05%, and the balance of Ni and inevitable impurities.
- 4-6. (Canceled)
7. (Previously Presented) A Ni-base superalloy according to Claim 2, wherein said weight % of Hf is 0.5 to 1%.
8. (Previously Presented) A Ni-base superalloy according to Claim 3, wherein said weight % of Hf is 0.5 to 1%.
9. (Previously Presented) A gas turbine component wherein it is manufactured by using said Ni-base superalloy as defined in Claim 2.
10. (Previously Presented) A gas turbine component wherein it is manufactured by using said Ni-base superalloy as defined in Claim 3.
11. (Canceled)
12. (Previously Presented) A gas turbine component wherein it is manufactured by using said Ni-base superalloy as defined in Claim 7.

13. (Previously Presented) A gas turbine component wherein it is manufactured by using said Ni-base superalloy as defined in Claim 8.

14. (Previously Presented) A gas turbine component according to Claim 9, wherein said gas turbine component is manufactured by a directional solidification casting method.

15. (Previously Presented) A gas turbine component according to Claim 10, wherein said gas turbine component is manufactured by a directional solidification casting method.

16. (Canceled)

17. (Previously Presented) A gas turbine component according to Claim 12, wherein said gas turbine component is manufactured by a directional solidification casting method.

18. (Previously Presented) A gas turbine component according to Claim 13, wherein said gas turbine component is manufactured by a directional solidification casting method.